

Simulation and Optimization of a Lunar Plume-Surface Interaction Measurement System

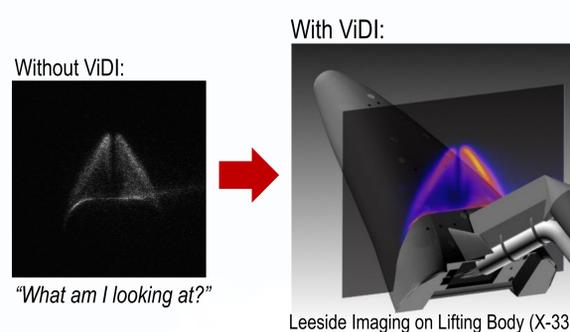


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Overview:

The Virtual Diagnostics Interface (ViDI) is applied for the analysis, simulation and design of a camera system that will perform photogrammetry during descent and post-landing on the surface of the Moon to assess plume-surface interaction.

What is ViDI?

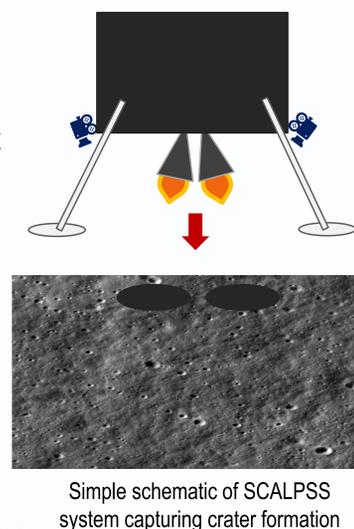


Virtual environment that combines 3D modeling, camera simulation and data visualization to plan and optimize diagnostic experiments. Typically used for wind tunnel experiments at NASA Langley Research Center.

What is SCALPSS?

Stereo Cameras for Lunar Plume Surface Studies

- What: a multiple camera photogrammetry system that will collect images of the lunar surface during descent and post landing.
- Objective: study the topic of plume-surface interaction on the lunar surface.
 - Surface Erosion
 - Regolith Ejection
 - Crater Formation
- Why: these phenomena can cause damage and health risks to mission infrastructure and crew on future manned missions to the Moon and Mars and have limited testing and simulation capabilities on Earth.



- Landing on the Moon in 2022
- Four camera system to image the crater formed by main engine thruster after landing.



- Landing on the Moon in 2023
- Six camera system to image the undisturbed regolith during descent and the craters formed by four pairs of attitude control thrusters after landing.

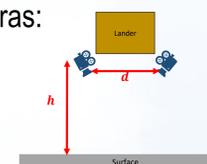
Simulation and Analysis:

- **Challenge: SCALPSS 1.1 aims to capture stereo imagery both before and after erosion of the regolith has occurred.**
 - At what altitude will erosion start to occur?
 - What is the necessary accuracy for the system to take measurements above the surface, and how can this be predicted?
 - How large of an area can the cameras capture and how long will their views be overlapping? What is our area of interest (AOI)?

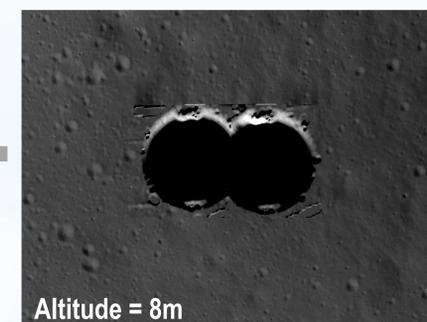
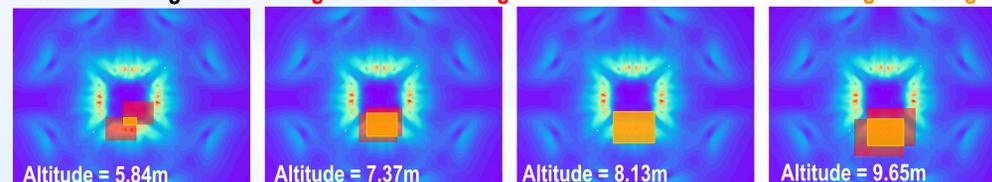
Long Focal Length Cameras (2) – capture images of partial AOI prior to erosion during descent

Development of empirical model for accuracy of a stereo pair of cameras:

$$Accuracy_{RMSE}(mm) = 0.611 * \left(\frac{3.37}{f}\right) \left(\frac{1.9}{d}\right) (h)^{1.79}$$

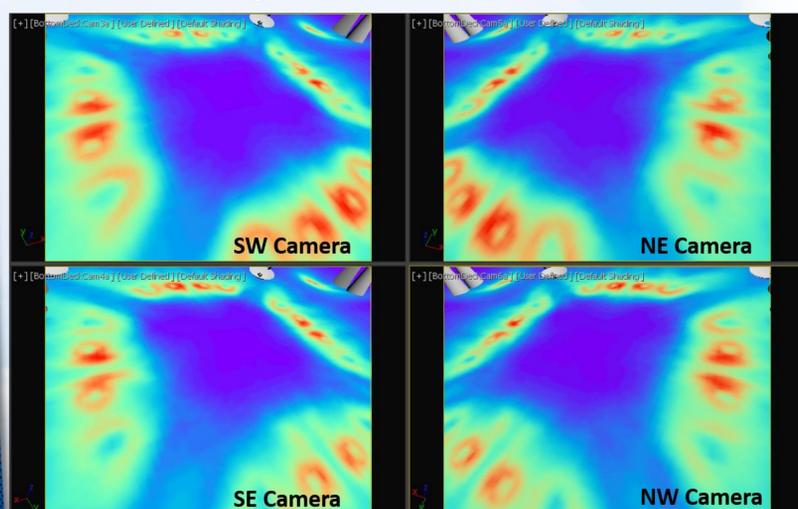


50mm focal length lens Single camera coverage in red Stereo camera coverage in orange



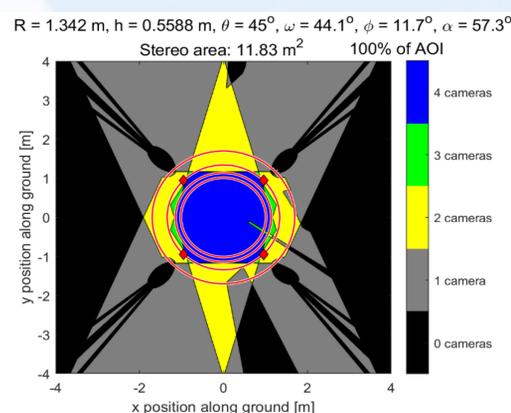
Simulated long focal length camera image of a pair of craters forming underneath an ACS thruster pair.

Short Focal Length Cameras (4) – capture images of full AOI after landing



Simulated camera views of the four short focal length cameras with the erosion CFD performed at NASA Marshall Space Flight Center.

Diagram of CFD-derived solutions for erosion of the lunar surface underneath the four pairs of thrusters with projected camera views at varying altitudes.

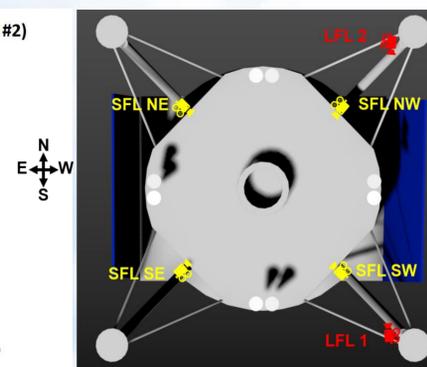
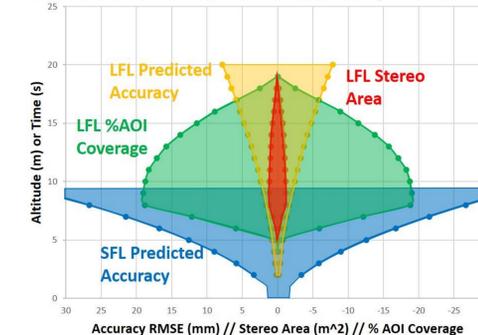


Stereo coverage and field of view plot for a selected four short focal length camera configuration.

Results:

Cohesive prediction of camera coverage, accuracy, and imagery during descent and post-landing informing an optimized camera configuration.

Descent Profile for Proposed Configurations (LFL #4 and SFL #2)



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